## **GUJARAT TECHNOLOGICAL UNIVERSITY**

**BE - SEMESTER-IV (NEW) EXAMINATION - SUMMER 2024** 

Subject Code:3140101 Date:03-0		<b>'-2024</b>		
S	ubjec	et Name:Aircraft Structures		
	Time:10:30 AM TO 01:00 PM Total Mar		ks:70	
Instructions:				
		<ol> <li>Attempt all questions.</li> <li>Make suitable assumptions wherever necessary.</li> </ol>		
		3. Figures to the right indicate full marks.		
	4	4. Simple and non-programmable scientific calculators are allowed.	MADEC	
			MARKS	
<b>Q.1</b>	(a)	What are the different loads acting on aircraft?	03	
	<b>(b)</b>	With neat sketch explain spar, ribs, stringers and skin functions.	04	
	(c)	What is V-N diagram? Explain its importance as a structure point of view.	07	
Q.2	(a)	What is the difference between Symmetrical Bending and Unsymmetrical Bending?	03	
	<b>(b)</b>	Explain reciprocal theorem.	04	
	(c)	Explain the energy method to calculate the buckling loads in columns.  OR	07	
	<b>(c)</b>	Explain the principal of least work for statically indeterminate structure.	07	
Q.3	(a)	Enlist different types of trusses.	03	
	<b>(b)</b>	Explain the role of bulkheads and longerons in detail.	04	
	(c)	Enlist various methods to find slope and deflection. Mention the assumptions required for deriving the differential equation.	07	
0.3	(.)	OR	0.2	
Q.3	(a) (b)	Explain how to find of degree of indeterminacy of structure.  Derive the strain energy equation for a member subjected to shear force.	03 04	
	(c)	Derive the strain energy equation for a member subjected to shear force.  Derive the equation for the shear flow of open section.	07	
Q.4	(a)	Define: Stress, Strain, Shear Modulus.	03	
		Write down the assumptions made in the theory of plane truss.	04	
	(c)	Explain the derivation for bending stress in unsymmetrical section.  OR	07	
Q.4	(a)	Enlist the different type of materials used in aircraft.	03	
	<b>(b)</b>	Explain theorem of virtual work and its applications.	04	
	(c)	State the assumptions and limitations of Euler's Theory of Column Buckling.	07	
Q.5	(a)	Define principle moment of inertia.	03	
	<b>(b)</b>	Explain monocoque fuselage structure.	04	
	(c)	Determine the value of slope at the free end of the cantilever beam with UDL of W and length L using Area-Moment method.	07	
o -		OR	0.2	
Q.5	(a)	Write down the difference between torsion of open and closed Sections.	03 04	
	(b) (c)	Explain semi monocoque fuselage structure. Explain strain and displacement relationships for open and single cell closed section thin-walled beams.	07	

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