

GUJARAT TECHNOLOGICAL UNIVERSITY**BE- SEMESTER-VI (NEW) EXAMINATION – WINTER 2024****Subject Code:3160113****Date:02-12-2024****Subject Name: Advance Aerodynamics****Time:02:30 PM TO 05:00 PM****Total Marks:70****Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.
4. Simple and non-programmable scientific calculators are allowed.
5. Use drawing instruments to make figures.

	MARKS
Q.1 (a) What do you mean by Aerothermodynamics?	03
(b) Define Mach number and Shockwave.	04
(c) Explain construction of subsonic open type wind tunnel with neat sketch.	07
Q.2 (a) Prove $C_p = 2\sin^2 \theta$ for Newtonian theory.	03
(b) What is the difference between supersonic and hypersonic wind tunnel?	04
(c) Explain Centrifugal force corrections to Newtonian theory with neat sketch.	07
OR	
(c) Explain Critical Mach number and Drag divergence mach number.	07
Q.3 (a) Enlist Application to supersonic airfoils.	03
(b) List out flow visualization techniques.	04
(c) Explain Solid blockage and wake blockage.	07
OR	
Q.3 (a) Define wind tunnel balances.	03
(b) Explain Hypersonic expansion wave relation.	07
(c) Explain Aerodynamic heating.	
Q.4 (a) Explain viscous retraction with sketch.	03
(b) Define The velocity potential equation.	04
(c) Why car is not fly at Subsoinc, Supersoinc and hypersonic speed?	07
OR	
Q.4 (a) Explain flow over an airfoil case for hypersonic case.	03
(b) With neat sketch explain thin shock layer.	04
(c) Define Supercritical Airfoil, Rayleigh flow and Fanno flow.	
Q.5 (a) What is The sound barrier and Area Rule.	03
(b) Explain Low density flow.	04
(c) Derive $L/D = C_{D\alpha}$ equation for flat plate using aerodynamic forces.	07
OR	
Q.5 (a) Define compressible flow and Incompressible flow.	03
(b) Write a note on “Prandtl-Glauert Compressibility correction”.	04
(c) Explain Rayleigh flow with sketch.	07
