

GUJARAT TECHNOLOGICAL UNIVERSITY**BE - SEMESTER-VI(NEW) EXAMINATION – WINTER 2022****Subject Code:3160113****Date:16-12-2022****Subject Name:Advance Aerodynamics****Time:02:30 PM TO 05:00 PM****Total Marks:70****Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.
4. Simple and non-programmable scientific calculators are allowed.

MARKS

- | | | |
|------------|--|-----------|
| Q.1 | (a) What do you mean by Aerothermodynamics? | 03 |
| | (b) Difference between subsonic and supersonic flow. | 04 |
| | (c) Define Hypersonic flow. Explain with neat sketch thin shock layer and entropy layer. | 07 |
| Q.2 | (a) Define Mach number and Shockwave. | 03 |
| | (b) Explain viscous retraction with sketch. | 04 |
| | (c) Write a note on Recapitulation with neat sketch. | 07 |
| | OR | |
| | (c) Explain shock expansion theory. | 07 |
| Q.3 | (a) Define shock angle and deflection angle. | 03 |
| | (b) Explain Low density flow. | 04 |
| | (c) Derive an equation on “Hypersonic shock relation in terms of hypersonic similarity parameters”. | 07 |
| | OR | |
| Q.3 | (a) Discuss in short “wave rider”. | 03 |
| | (b) Explain Newtonian flow with appropriate sketch. | 04 |
| | (c) Derive $L/D = C_o \tan \alpha$ equation for flat plate using aerodynamic forces. | 07 |
| Q.4 | (a) Define compressible flow and Incompressible flow. | 03 |
| | (b) Explain Modified Newtonian law with appropriate sketch. | 04 |
| | (c) Discuss on comparison between newtonian and exact results for the pressure coefficient on a sharp wedge and a sharp cone | 07 |
| | OR | |
| Q.4 | (a) Define critical mach number. | 03 |
| | (b) Discuss supersonic flow over a cone. | 04 |
| | (c) Write a note on Aerodynamic heating. | 07 |
| Q.5 | (a) Define Drag divergence mach number. | 03 |
| | (b) Write a note on “Prandtl-Glauert Compressibility correction”. | 04 |
| | (c) Derive velocity potential equation with sketch. | 07 |
| | OR | |
| Q.5 | (a) What do you mean by supersonic airfoil drag? | 03 |
| | (b) Derive “Linearized supersonic pressure coefficient formula” | 04 |
| | (c) Explain Rayleigh flow with sketch. | 07 |
