

**GUJARAT TECHNOLOGICAL UNIVERSITY**

**BE - SEMESTER-VII (NEW) EXAMINATION – SUMMER 2024**

**Subject Code:3170110**

**Date:22-05-2024**

**Subject Name:Introduction to Aeroelasticity**

**Time:02:30 PM TO 05:00 PM**

**Total Marks:70**

**Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.
4. Simple and non-programmable scientific calculators are allowed.

**MARKS**

<b>Q.1</b>	(a) What is the role of Aeroelasticity in aviation field? Explain in short.	<b>03</b>
	(b) Difference between 2-D airfoil and wing.	<b>04</b>
	(c) Write a short note on “Deformation of Structures”.	<b>07</b>
<b>Q.2</b>	(a) Classify Aeroelasticity.	<b>03</b>
	(b) Define Static aeroelasticity.	<b>04</b>
	(c) What is Energy Method? Explain	<b>07</b>
<b>OR</b>		
	(c) Discuss on Influence Coefficients.	<b>07</b>
<b>Q.3</b>	(a) Discuss shortly Aileron reversal.	<b>03</b>
	(b) Define Dynamic aeroelasticity.	<b>04</b>
	(c) Discuss on “U-g Method”.	<b>07</b>
<b>OR</b>		
<b>Q.3</b>	(a) Why flutter form? Shortly explain	<b>03</b>
	(b) What is the difference between Flutter and Vibration?	<b>04</b>
	(c) How to find Lift Distribution for the Steady Roll Case?	<b>07</b>
<b>Q.4</b>	(a) Define Swept wing.	<b>03</b>
	(b) What is finite state model?	<b>04</b>
	(c) Discuss on Flutter model of 2-D Airfoil.	<b>07</b>
<b>OR</b>		
<b>Q.4</b>	(a) Define straight wing.	<b>03</b>
	(b) Difference between 2D and 3D supersonic flow over body.	<b>04</b>
	(c) Explain Kernal Function Approach.	<b>07</b>
<b>Q.5</b>	(a) Define Subsonic flow.	<b>03</b>
	(b) Discuss on “Aerodynamic lift for a Harmonically oscillating Aerofoil”.	<b>04</b>
	(c) Write a note on Theodorsen Theory.	<b>07</b>
<b>OR</b>		
<b>Q.5</b>	(a) Define Supersonic flow.	<b>03</b>
	(b) How to minimize flutter problem in wing?	<b>04</b>
	(c) Discuss p-k method.	<b>07</b>

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