

**GUJARAT TECHNOLOGICAL UNIVERSITY**

**BE - SEMESTER-VII EXAMINATION – SUMMER 2025**

**Subject Code:3170114**

**Date:08-05-2025**

**Subject Name:Space Flight Mechanics**

**Time:02:30 PM TO 05:00 PM**

**Total Marks:70**

**Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.
4. Simple and non-programmable scientific calculators are allowed.

		MARKS
<b>Q.1</b>	(a) What is Space? Is there gravity in space?	<b>03</b>
	(b) What is zero potential energy configuration?	<b>04</b>
	(c) Derive orbit equation.	<b>07</b>
<b>Q.2</b>	(a) Is there gravity in space? Justify.	<b>03</b>
	(b) Write difference between Elliptical and Circular orbit.	<b>04</b>
	(c) Write a short note Escape Velocity.	<b>07</b>
	<b>OR</b>	
	(c) Explain The Two body problem.	<b>07</b>
<b>Q.3</b>	(a) Derive the external force acting on rigid body by using Newton's second law of motion.	<b>03</b>
	(b) State and Prove Kepler's 3rd law.	<b>04</b>
	(c) Briefly classify Space vehicles.	<b>07</b>
	<b>OR</b>	
<b>Q.3</b>	(a) Explain Mechanics of Circular Orbit.	<b>03</b>
	(b) Explain Gravitational potential energy.	<b>04</b>
	(c) Explain Hohmann transfer ellipse.	<b>07</b>
<b>Q.4</b>	(a) Write a short note on Rigid body.	<b>03</b>
	(b) Write a note on skip reentry dynamics.	<b>04</b>
	(c) Explain different types of entry paths.	<b>07</b>
	<b>OR</b>	
<b>Q.4</b>	(a) Explain Entry heating.	<b>03</b>
	(b) Explain steep ballistic reentry.	<b>04</b>
	(c) With neat sketches explain different space trajectories and its physical significance.	<b>07</b>
<b>Q.5</b>	(a) What is gyrostat?	<b>03</b>
	(b) Explain non spinning satellite of attitude control.	<b>04</b>
	(c) Establish a relation between Impulse and change in momentum.	<b>07</b>
	<b>OR</b>	
<b>Q.5</b>	(a) Explain Escape velocity. Calculate the same for an object escaping from the earth's surface.	<b>03</b>
	(b) What are the functions of (a) Heat Shield & (b) Back Shell?	<b>04</b>
	(c) Define Entry heating. Derive an expression for aerodynamic heating rate.	<b>07</b>

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